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Problem Statement

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- Delamination is a common failure on polymer matrix composites (PMC)
	- Sources include manufacturing defects and inservice or maintenance damage
- Resin injection is a standard repair technique but NOT considered strengthrestoring
	- Difficulty in characterizing and removing contaminants in small cracks
	- Multiple drilling attempts and heat application required to inject currently approved resins
- Benefit of resin injection: faster and cheaper compared to patch repairs

Injection Repair

Problem Statement, Cont.

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- Per aircraft maintenance and repair manuals, resin injection is used to repair delaminated composite materials using laminating resins Henkel EA9396 and EA956.
- Filling small cracks $($ \leq 1 micron) is difficult due to the resins' viscosity (~3500 cP at 77°F)
	- Delaminated sections are rarely filled completely without heat application and multiple attempts
	- Current injection repair process are considered non-structural due to repetitive drilling and incomplete crack fill
	- Previous research diluted EA9396 with 10% and 20% w/w acetone to achieve lower viscosity at the expense of increased porosity [1]
- FRCSW is investigating epoxies that:
	- (1) initially meet a viscosity of <650 cP, recommended by A.J. Russell (Canadian Defense Research Establishment Pacific) [2]
	- (2) comply with the in-service temperature and humidity conditions of multiple aircraft platforms.

- Milestone 1: Identify potential resin candidates that satisfy the following criteria:
	- Viscosity < 650 cP
	- Tg (glass transition temperature) > 250°F
	- Pot life is ≥ 20 minutes
- Milestone 2: Characterize resin mechanical and chemical properties for convergence upon

2-3 resins

- Milestone 3: Test Mode II interlaminar fracture toughness of the selected epoxies to compare to baseline unidirectional laminate values
- Milestone 4: Inject resins into laboratory-impacted coupons. Perform NDI and mechanical testing to converge upon final 1-2 resins that will be used for multiple platforms in the field

Milestone 1: Epoxy Identification

Potential Epoxy Candidates

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Table 1: Potential Off-the-Shelf Products

Table 2: Potential EA9396 Dilutions

- Potential resin candidates need to satisfy the following criteria
	- Viscosity < 650 cP
	- Tg (glass transition temperature) > 250°F
	- Pot life is ≥ 20 minutes
	- Procurable (no excessive lead times or DOD-specific difficulties)
- Diluted EA9396 included for consideration due to all materials being readily available

Milestone 2: Epoxy Characterization

Viscometer Testing

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Brookfield Viscosities of Various Epoxies

• Evaluation of epoxy viscosity conducted with Brookfield Viscometer

- Reading were taken over the span of 10 minutes after time of mix
- Ambient temperature at 21°C versus the 25°C condition stated in TDS
	- Results are more viscous than values stated in datasheets, also more conservative and representative of actual conditions
- Results
	- SR6400 and EA9396-15% acetone display the lowest viscosity (200 300 cP)
	- Supreme 112 and Araldite well above the 650 cP of AJ Russell's epoxy (red line); IR6030 at the threshold
	- Not pictured: initial viscosity for 10% MIBK is 833.3 cP

Single Lap Shear

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Low Viscosity Resin Single Lap Shear Strength

■ Conditioned, Elevated Temperature ■ Dry, Room Temperature

- Conducted single lap shear (SLS) testing for dry and heat and humidity-conditioned samples
- Conditioned samples were in humidity cabinet at 140°F and 90% RH until moisture equilibrium achieved
	- Samples soaked for 10 minutes at 250°F immediately prior to testing
- Results
	- EA9396 variants and Araldite 5052 demonstrated the largest RT SLS and the most extreme change with hotwet samples
	- Kaneka resins displayed the lowest deviation between the hot-wet and dry SLS
	- All coupons failed cohesively

DSC Testing

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EA9396-15% Acetone- uncured sample example

DSC sample containers for EA9396-15% acetone (left) and EA9396 (right) post testing

- Using DSC to determine the degree of cure per the TDS-recommended cure cycles
	- Heating rate at 20°C/min
- Samples of known mass are heated and changes in heat capacity are tracked as a function of time
- Residual exotherm of cured and uncured samples are compared to get degree of cure
- Volatility of EA9396 diluted with acetone noted in epoxy egress from DSC container

10% MIBK EA9396 Resin Puck

Purpose: characterize the bubble generation from diluting EA9396 with different solvents

– Excess bubbles could become trapped in small cracks from delamination

0.010in

– Cure 10% and 20% MIBK after 24 hr set time at room temp in containers with lid closed to simulate "worst-case" (arduous path for resins to outgas)

20% MIBK, x 20 magnification

– Pucks" cut in half; cross-section polished and examined under microscope

Results

- 20% MIBK showed surface discoloration post cure and large voids at surface
	- Solvent concentration is rejected as a viable option
- 10% MIBK puck shows very sporadic voids- fairly homogeneous surface
- MIBK as a diluent was eliminated due to volatility and 10% w/w only reduced viscosity to 833 cP

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Porosity Studies-All Potential Epoxies

- Used Keyence VHX-7000 digital microscope to analyze porosity generated from hand-mixing and curing epoxies
	- Areal porosity calculated using area of bubbles (red) with respect to specified area of coupon
- Micrographs taken with x20 magnification to yield more global view of coupon (i.e. EA9396 shown below) **Notes**
	- EA9396 variants showed the largest areal porosity
		- Mixing Parts A and B components that are very different in viscosity requires more motion for homogenous solution

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• High viscosity of Part A also seems to result in large amount of bubbles induced • Observed comparatively low porosity in remaining epoxies

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DMA Testing

- Stress applied to heated sample in 3-pt bend configuration
- Material response is measured; loss in elasticity is correlated to Tg
- \cdot Test parameters and results interpretation dictated by ASTM 7028
- 3 samples were tested per epoxy
- **Results**
	- Both Kaneka resins exhibit average storage modulus Tg that is larger than expected Super Hornet service temperature (250°F)
	- All epoxies have comparatively larger Tg versus the currently-used EA9396

Decision Matrix Scoring Criteria

- Created a decision matrix (outlined on next slide) to converge upon 1-2 candidate epoxies to continue to evaluation process (repair strength restoration)
- Categories that epoxies' were evaluated upon outlined in table above
	- Epoxies were evaluated relative to each other- cut-off for each "score" reflects this
	- Epoxies with the largest total "score" will be used for further evaluation with ENF and CAI

Decision Matrix

Milestone 3: Fracture Toughness Testing (In Progress)/ Hot-Wet Tg Testing

End-Notched Flexure (ENF) Test

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FIG. 1 ENF Test Fixture and Specimen Nomenclature

- IM7/977-3 panels created with PTFE insert to simulate delamination
- Coupons are loaded under Mode II shear to propagate crack and measure fracture load
	- Fracture toughness is calculated as function of fracture load and geometric effects

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Cleaning and Resin Injection Process

- Small cracks within delaminated components can be filled with common contaminants (hydraulic fluid, jet fuel, etc.)
	- Contaminants would compromise bonding of injected resin to c/ep coupon
- ENF coupons were injected with hydraulic fluid to simulate worst-case scenario
- Cleaning process involving acetone flush followed by nitrogen purge and atmospheric plasma surface treatment developed to reduce contaminant volume [2]
	- Plasma treatment done with Surfx Atomflo 500L
	- Reduction of HF elements were tracked using Hiden Quantitative Gas Analyzer (mass spectrometer)
- \sim 15 mL repair resin injected into coupon immediately post cleaning

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- Pulse-echo ultrasonic scans with 10 MHz transducer and water as couplant
- Edges of repair region tend to have pockets of under filled areas or voids (white boxes)
- Larger viscosity of IR6030 yields more effort in injection and filling of cracks

Kaneka SR6400 $^{\sim}$ 250 cP

Kaneka IR6030 $~\sim$ 650 cP

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Work in progress

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Elevated Temperature-Wet Tg

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Elevated Temperature **Room Temperature**

Average Tg from DMA Storage Modulus Curves 300 Temperature (°F) 250 Temperature (°F) 200 150 100 50 0 IR6030 SR6400 EA9396 Elevated Temperature **Room Temperature**

Average Tg from DMA Tan δ Curves

Elevated Temperature **Room Temperature**

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Path Forward

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- Milestone 3: Fracture Toughness Testing
	- Continue ENF testing to determine repaired coupons' fracture toughness
	- Analyze fracture surfaces to discern failure modes
	- Develop elevated-temperature test setup
- Milestone 4: NDI, Mechanical Testing of Resin-Injected Impacted Coupons
	- Inject repair resins into laboratory-impacted coupons
	- Perform pulse-echo UT to determine percentage fill
	- Compression-After-Impact (CAI) testing to be performed at PAX to determine strength restoration from repair
	- Converge upon the repair resin of choice

• Implementation

- Present results to platform FSTs
- Execute additional testing and/or perform demonstration on actual component as identified by FST

[1] Russell, A.J. and Bowers, C.P., "Resin Requirements for Successful Repair of Delaminations", Proceedings of the 36th SAMPE Symposium, 15-18 April 1991, San Diego, CA.

[2] Massey, J. T. (2023). Novel Quantitative Composite Delamination Injection Repair Procedure and Characterization for Strength Restoration. *UC San Diego*. ProQuest ID: Massey_ucsd_0033D_22445. Merritt ID: ark:/13030/m5m99dk2. Retrieved from https://escholarship.org/uc/item/17n8b3vz

[3] ASTM Standard D7905, 2019,"Standard Test Method for Determination of the Mode II Interlaminar Fracture Toughness of Unidirectional Fiber-Reinforced Polymer Matrix Composites," ASTM International, West Conshohocken, PA, 2023

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